



36th Annual Gun & Ammunition Symposium & Exhibition



April 9-12, 2001
San Diego, CA

Gary T. Phillips
Program Manager

ALOR (Artillery Launched Observation Rounds)





April 10, 2001

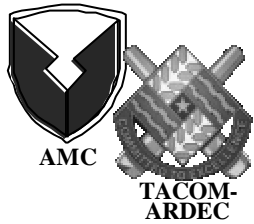
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Outline

- **ALOR DTO & Related Programs**
- **Operation & Employment of ALOR**
- **Current ALOR Program Goals**
- **ALOR Vehicle Technologies**
- **ALOR Progress to Date**



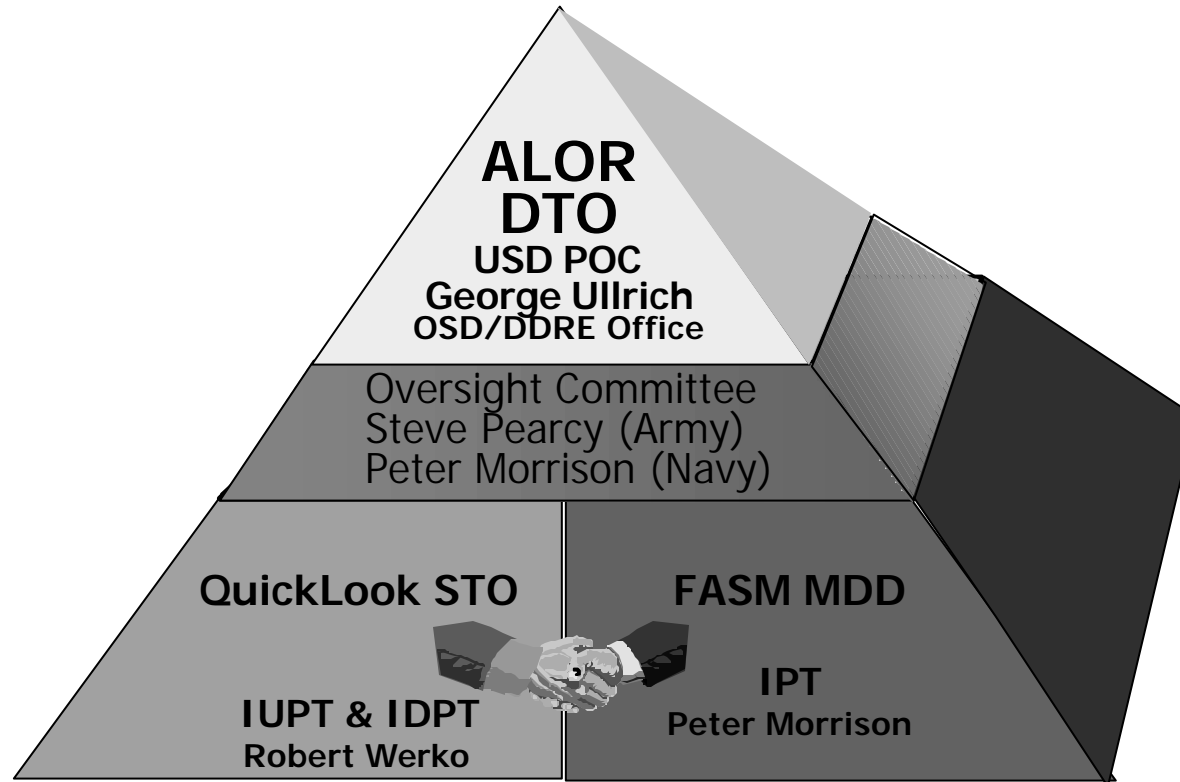


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ALOR DTO



Combines FASM & QuickLook Munitions Development



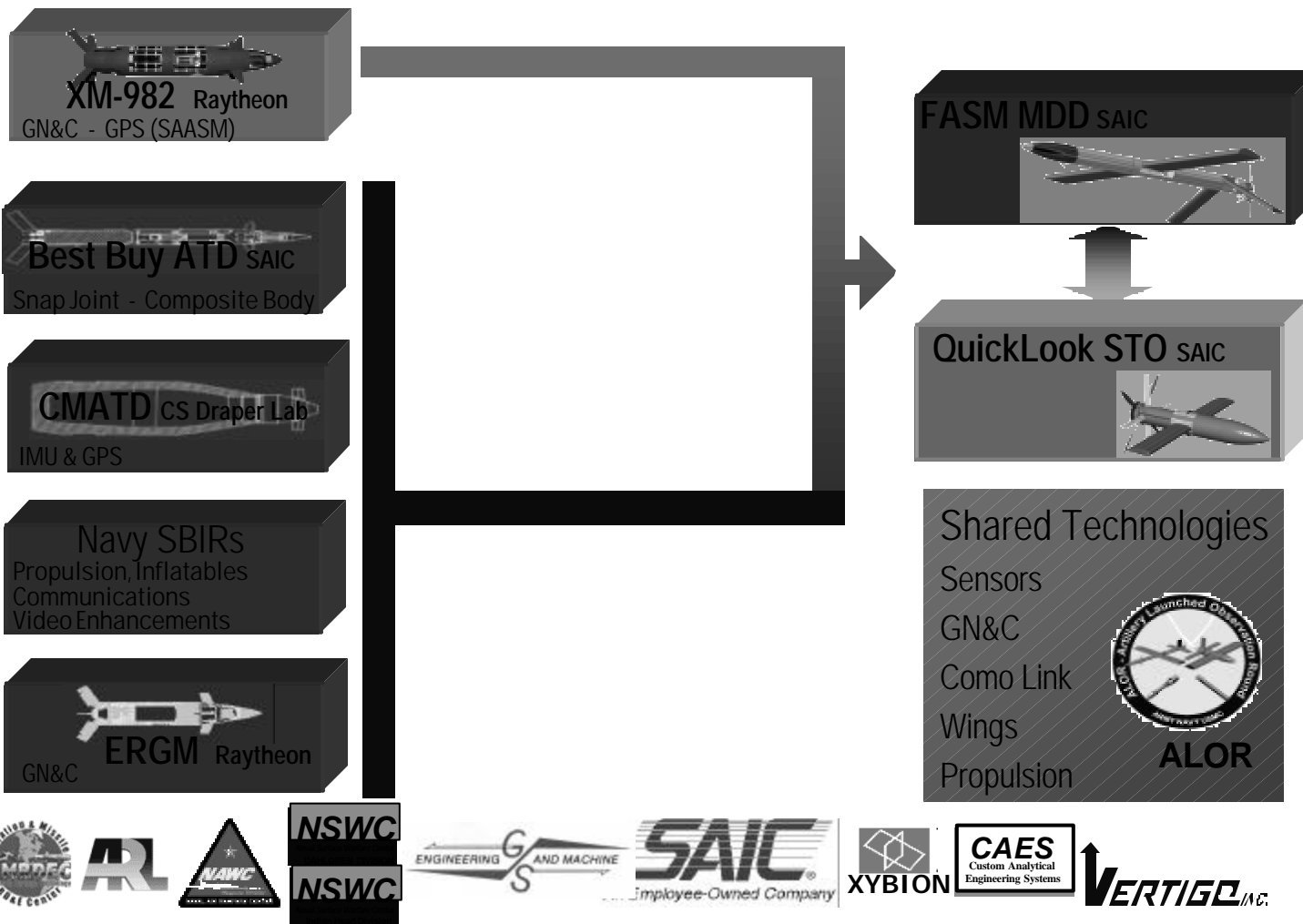


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FASM/QuickLook Leverage Navy & Army Programs



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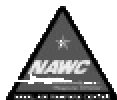
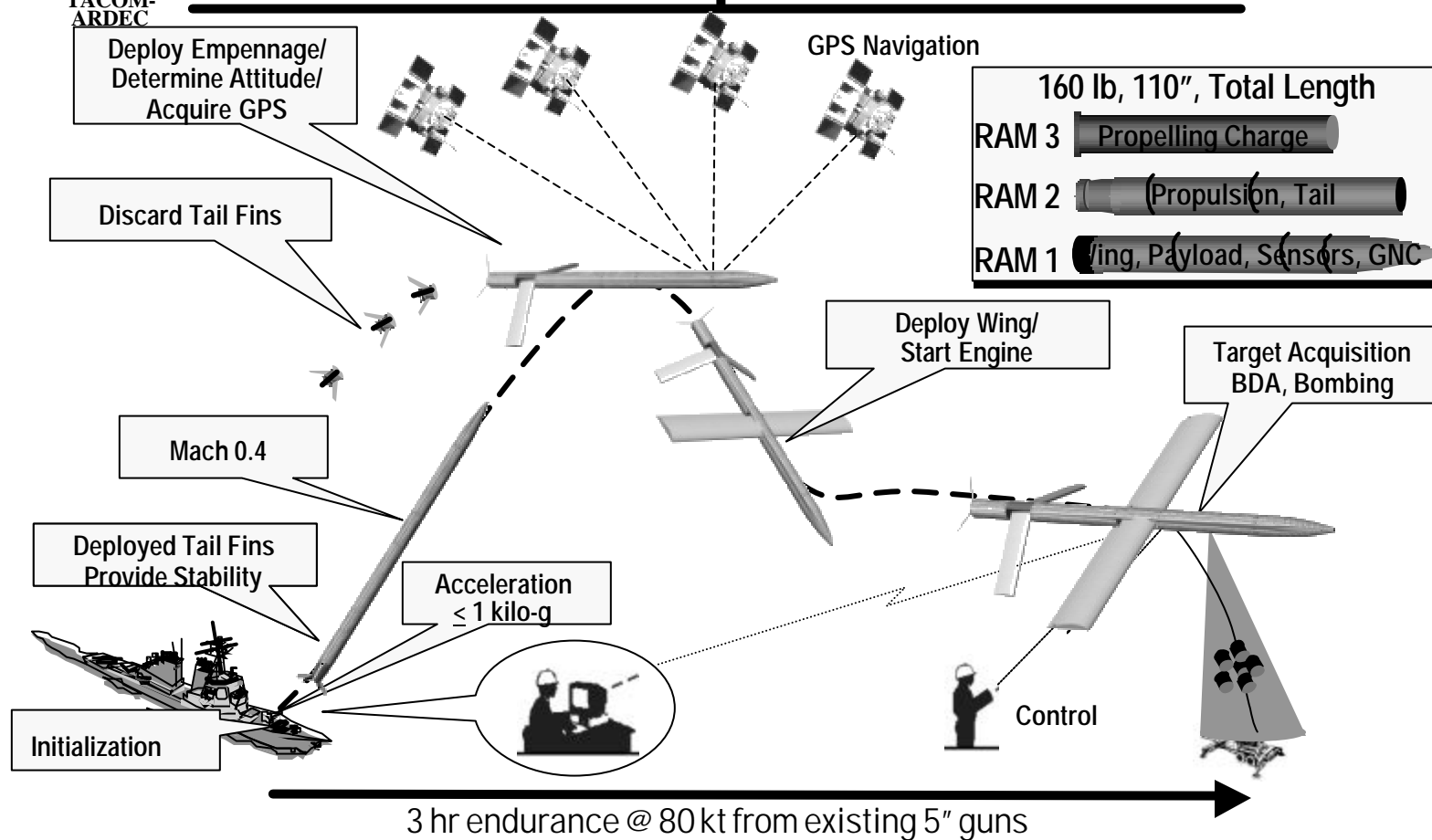


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FASM Operation





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FASM Concept of Employment

FASM provides an organic overhead com relay

Communications

FASM gives Marines <2 min response @ 100 nm for 2 hours

Flyout

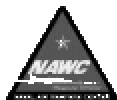
Weapon Delivery

Target Acquisition BDA

Gun Launch

FASM provides organic, autonomous, responsive targeting/BDA

FASM decreases TLE reducing # of ERGMS required to achieve high P_k



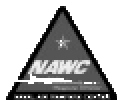
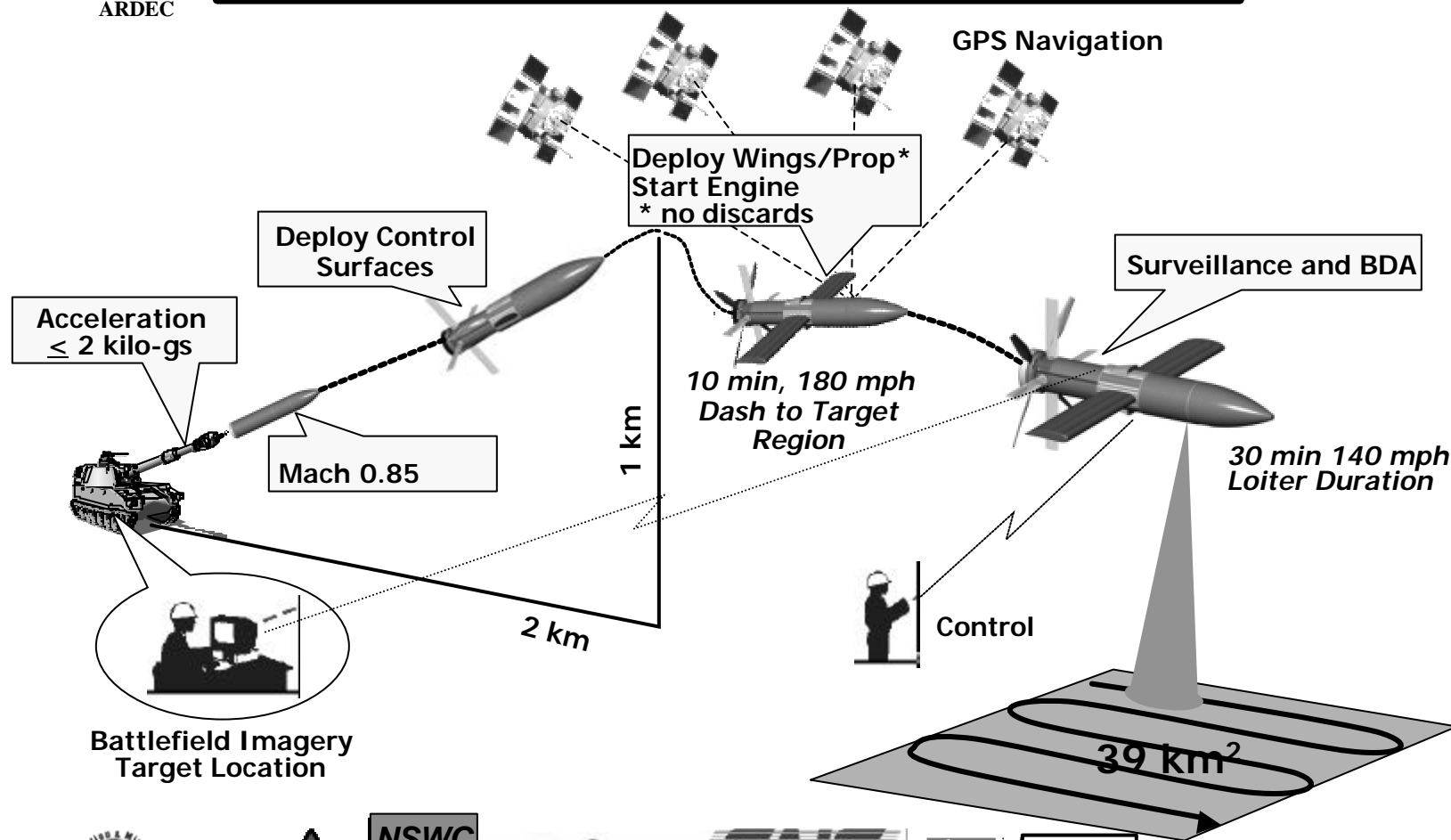


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QuickLook Operation



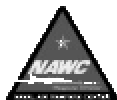
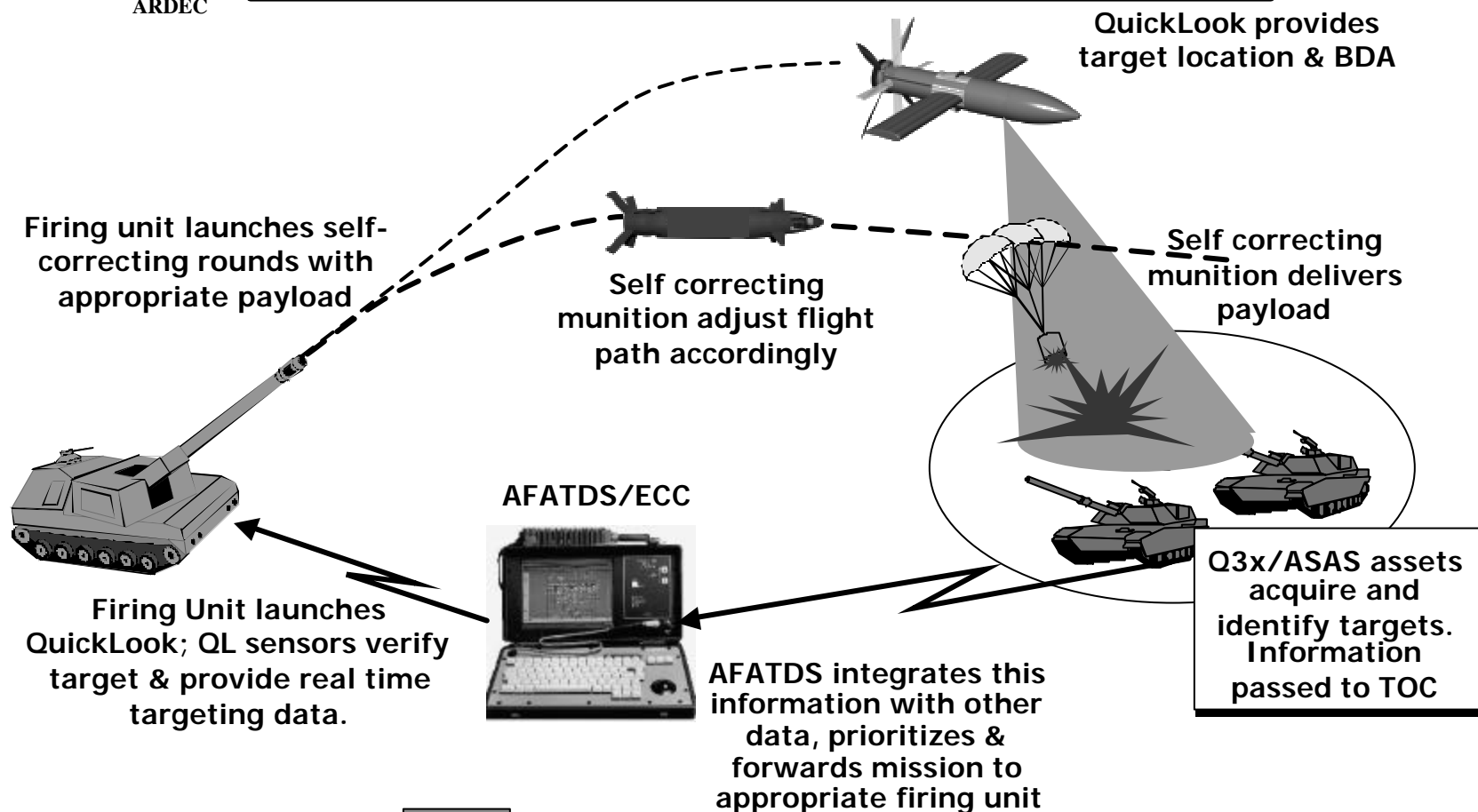


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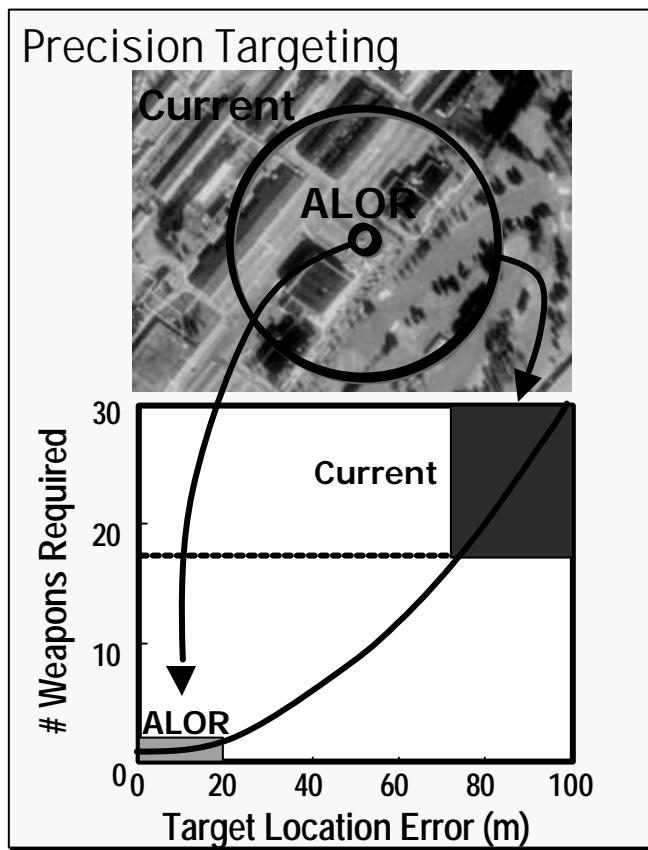
QuickLook Concept of Employment



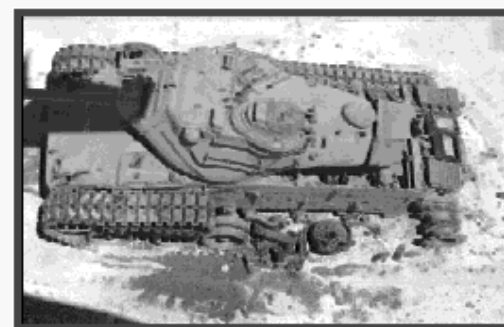
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Projected Impact on Battle

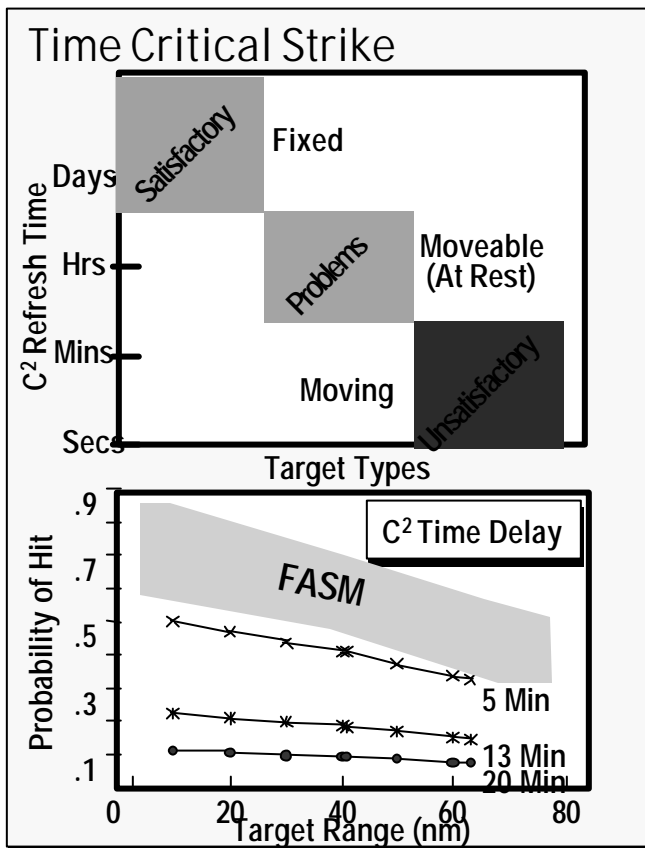


Battle Damage Assessment



- Debris
- Smoke
- Fire

Projected Impact on Battle



OMFTS



- Survey helicopter approach/retirement routes
- Survey HLZ
- Suppression in area around HLZ
- Protection during embarkation



- Survey enemy approach routes
- Act as a communication link
- Provide Suppressive Fire
- Provide Protective Fire



- BDA
- Survey approach / retirement Routes
- Act as a communication link
- Provide Suppressive Fire
- Provide Protective Fire



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FASM Marine Defense Demonstration Program

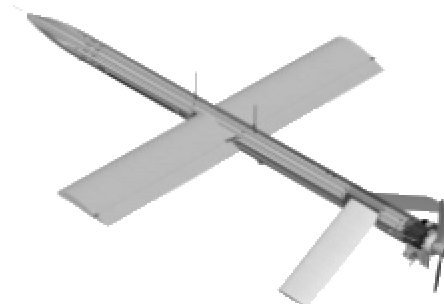


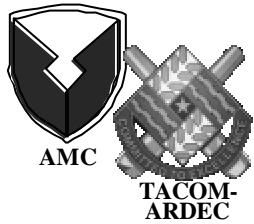
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The FASM MDD will demonstrate the following basic requirements:

- Ballistic configuration transition to stable cruise flight
- Cruise/Loiter endurance of more than 3 hours
- MIL tactical targeting using live imagery comm link
- Bomb drops with 5-m precision relative to airframe
- Battle damage assessment
- Autonomous flight with in-flight ground control redirect



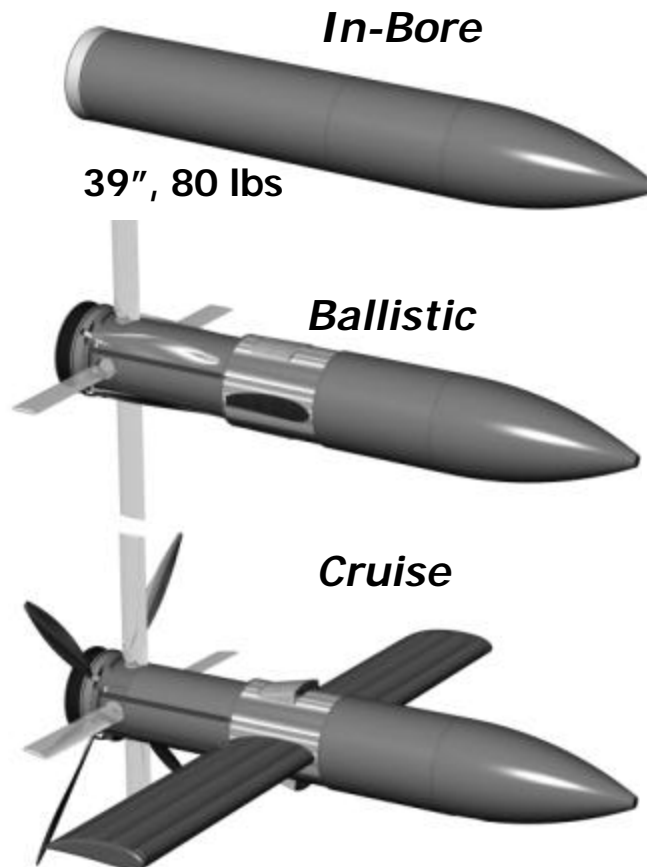


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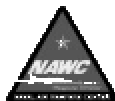


QuickLook STO Program



The QuickLook STO program will demonstrate the following as basic requirements:

- Transition from ballistic configuration to stable cruise flight.
- Loiter endurance of more than 30 minutes.
- Targeting with under 50-m CEP
- Battle damage assessment using live imagery communications linkage
- Autonomous flight with in-flight ground control redirect



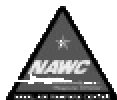
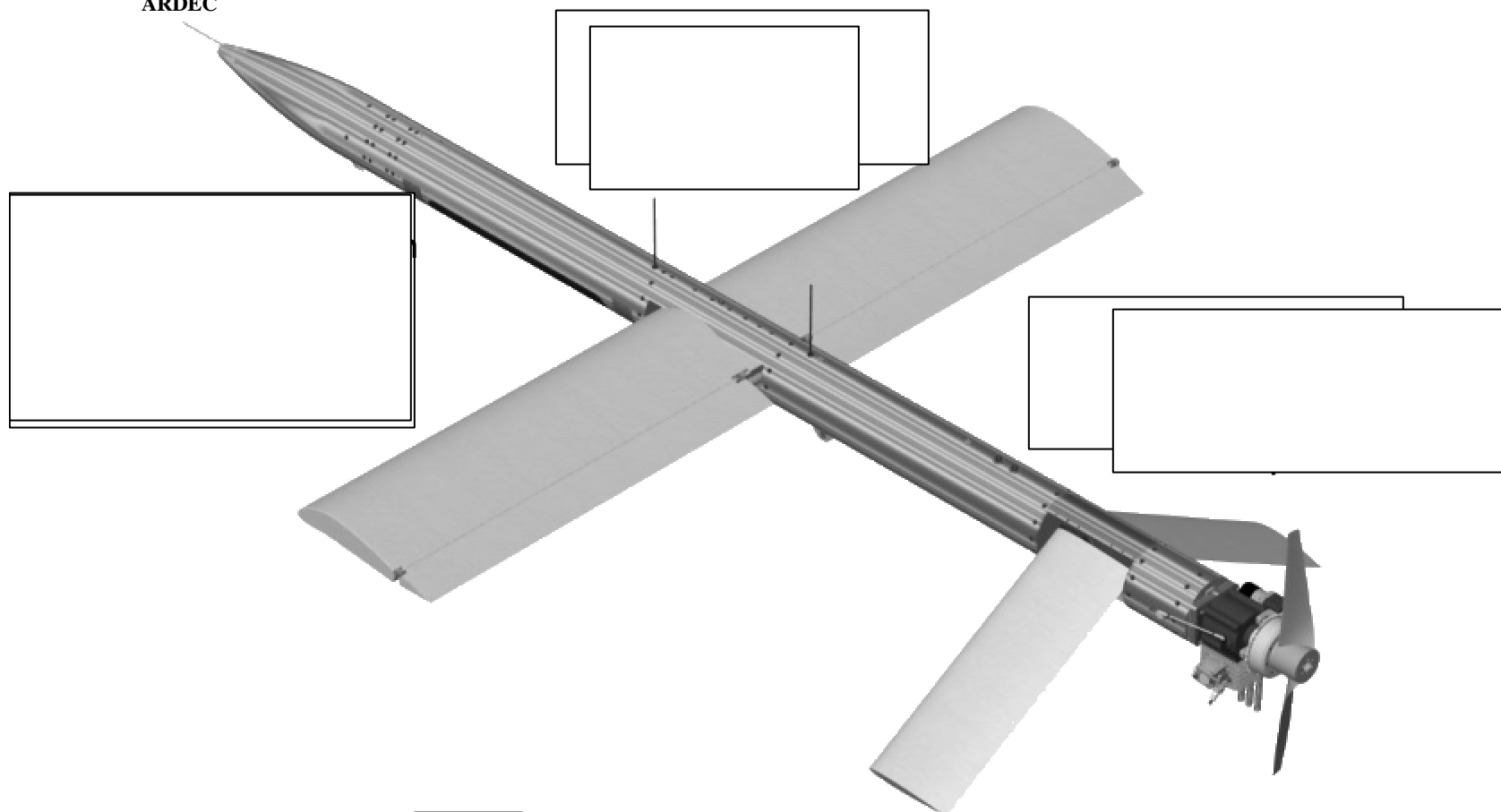


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Vehicle Technologies

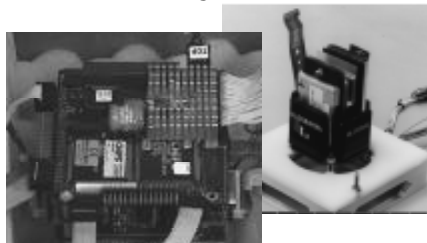




Tech Base



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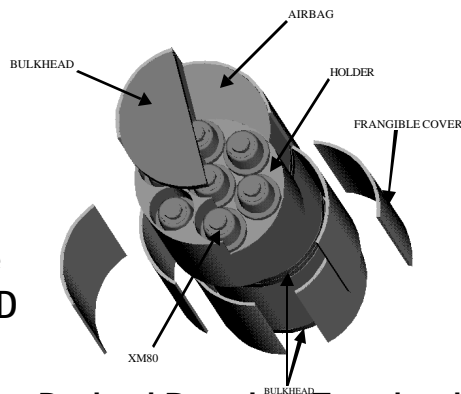
Flight Computer based on Vigilante
GNC relies on ERGM, XM982, CMATD



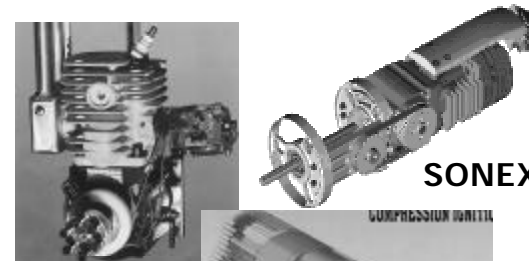
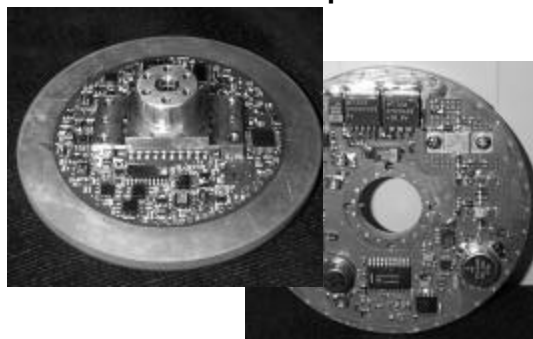
Inflatables based on GLOV SBIR



Advanced Composites based on Best Buy
Gun Hardened Camera & Transmitter
Based On Air Force BDA Camera Unit



Payload Based on Tomahawk
109-D Dispenser



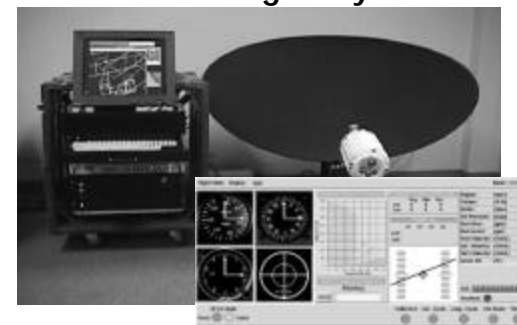
Brison

SONEX

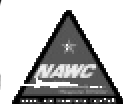


GSE

Propulsion Based on COTS, SBIR
& New Engine Systems



Ground Station and Flight Software Based
on Vigilante Autonomous Helicopter



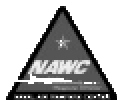
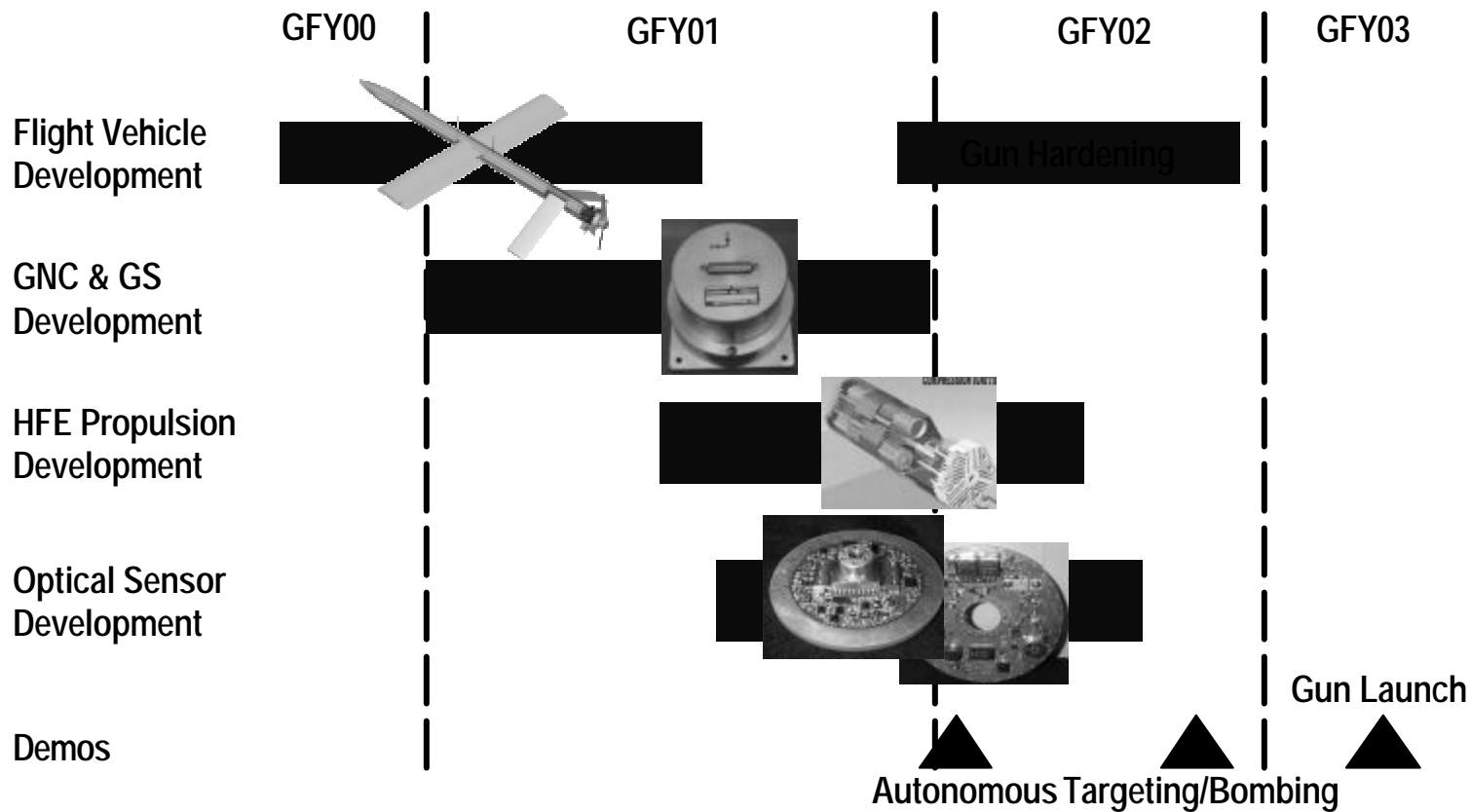


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Schedule





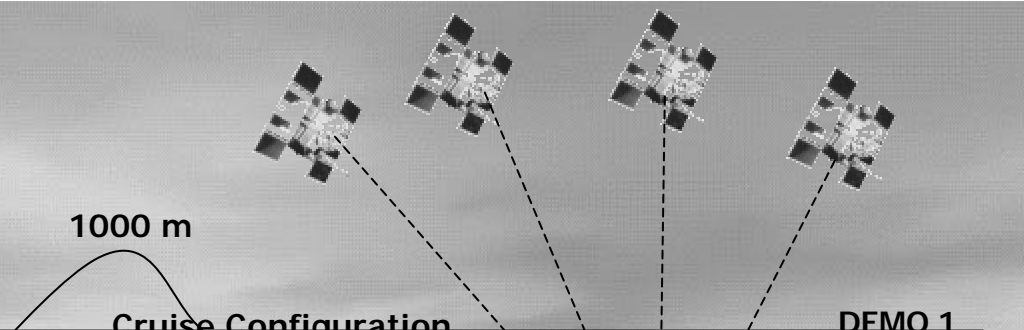
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Program Demos

Event	Time - min	Speed - mph
Launch	0	450
Apogee	0.2	225
Transition	0.2 – 0.5	245
Dash	0.5 – 10	185
Loiter	10 – 40	140

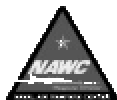


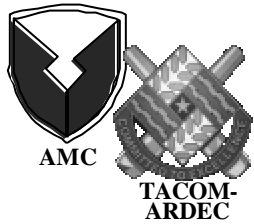
ALOR will demonstrate program objectives by combining results from drop & gun launch tests

Demonstrate
Gun Launch & Transform to
Cruise Configuration

DEMO 1
Demonstrate video & data TM to GS

DEMO 1
Demonstrate Target
Location Accuracy





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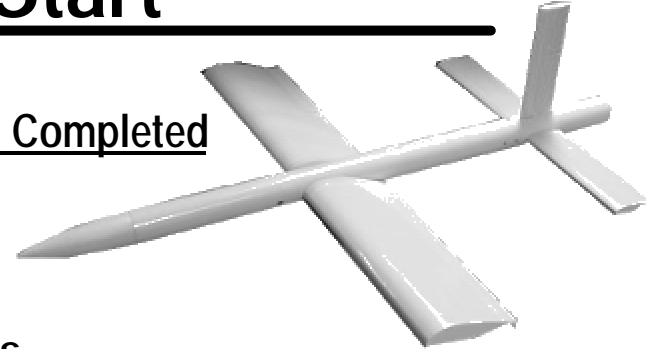
Strong Progress Since Program Start



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Full Scale, Non-Powered R/C Flight Tests Completed

- Stability and Maneuverability
- "Tail Sizing"
- Characterization complete June-00

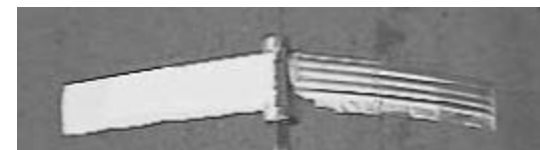
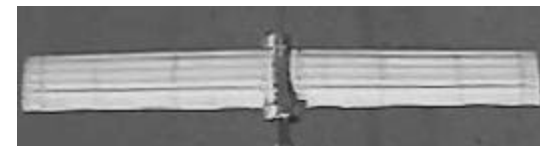


Engine Tests

- COTS engine characterization complete September-00
- HFE axial engine bench top demo complete December-00

Wing Deployment Tests (GLOV SBIR)

- Successful deployment @ 80 mph
- Wing deploys symmetrically with minimal rebound
- Wing spar pressure: 150 psi





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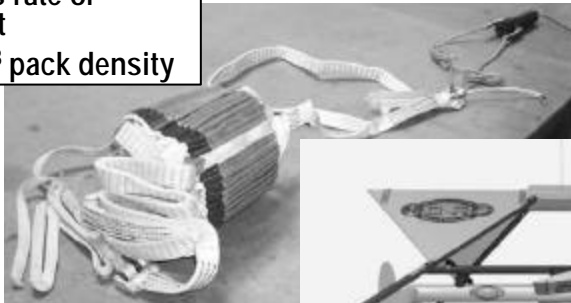
Strong Progress Since Program Start



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Parachute Specs

- 13.5 ft diameter
- 90% hemi
- 19.5 fps rate of descent
- 30 lb/ft³ pack density



Rocket Deployed Parachute Recovery System

- Chute housed in payload section
- Flight Tested in December-00

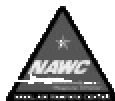
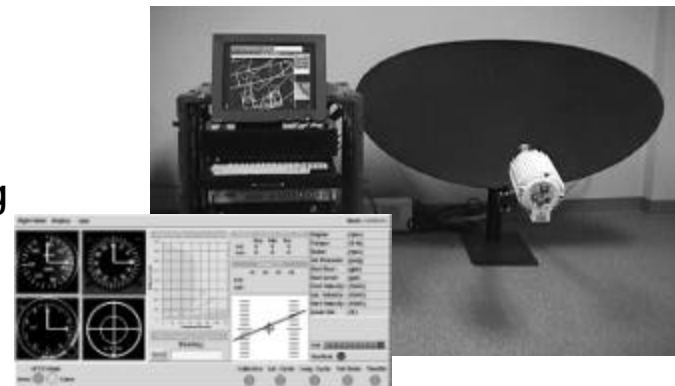


Launch Cradle

- Helicopter tow
- Flight Tested in November-00

Flight Software/Hardware & Ground Station

- R/C flight software/hardware and diagnostics complete December-00
- Executive Flight software architecture defined and coding complete July-01
- Vigilante Ground Station System Mod complete April-01
- Integration of flight computer hardware complete May-01





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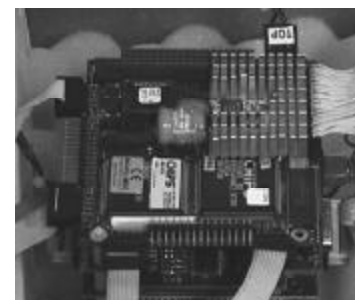
Strong Progress Since Program Start



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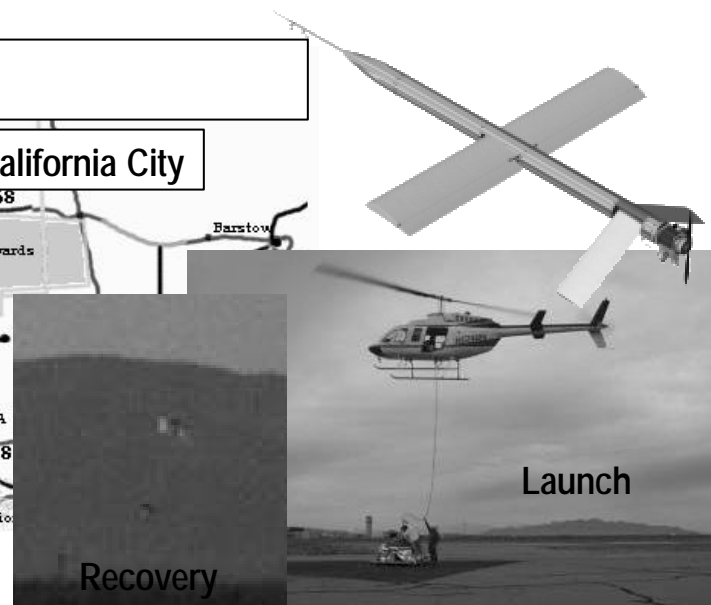
GN&C

- Hardware evaluation to begin in April-01
 - Rockwell NavStrike II GPS delivery May-01
 - Honeywell RLG consignment delivery March-01
 - Testing of PC-104 Flight Computer hardware underway
 - Mod 0 guidance law under development and complete by March-01
 - Mod 0 autopilot under development and complete by March-01

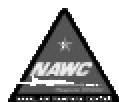


Full-Scale, Powered R/C Flight Vehicle

- Design Complete (R/C controls, air data & diagnostic sensors, on-board data acquisition, 2.4 GHz transceiver, COTs engine, CCD camera)
- Fabrication completed November-00
- Integration completed December-00
- Ground test completed by January-01
- Initial Flight Test February-01
- Visible sensor Flight Test February-01



Recovery





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Summary

Artillery Launched loitering munitions provide the following features:

- Low cost
- Expendable
- Quickly deployable
- Organic to early entry forces
- Sensor-to-shooter field artillery responsiveness
- Time Critical Strike (FASM, only)

